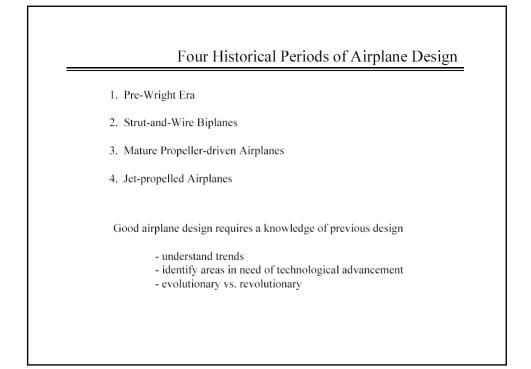
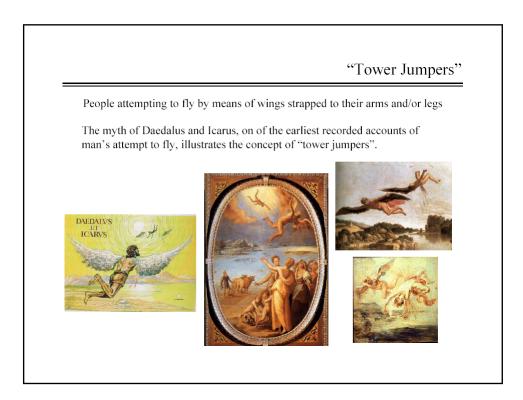
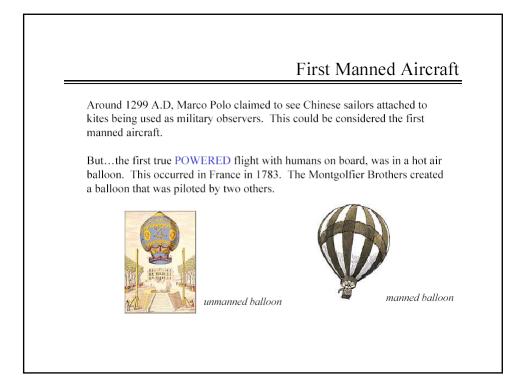
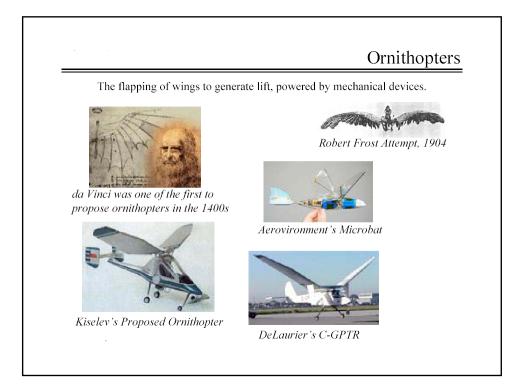


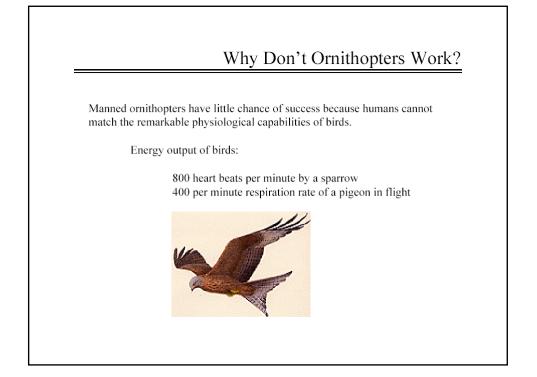
Detailed Outline	
Introduction	
Standard Atmosphere	
Basic Aerodynamic Concepts	(Chapter 2)
Airfoil Theory	
Wing Theory	
Aircraft Drag Estimation	
Aircraft Propulsion	(Chapter 3)
Thrust, Power and SaFC	
Turboprop Engines	
Turbofan Engines Equations of Motion	(Chapter 4)
Steady Flight	(Chapter 5)
Steady, level flight	(Onapter 5)
Climb and Drift-down	
Range	
Endurance	
Accelerated Flight	(Chapter 6)
Maneuvering	(
V-n Diagram	
Take-off	
Landing	

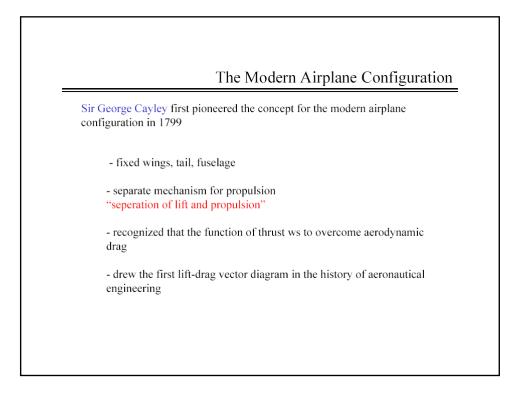


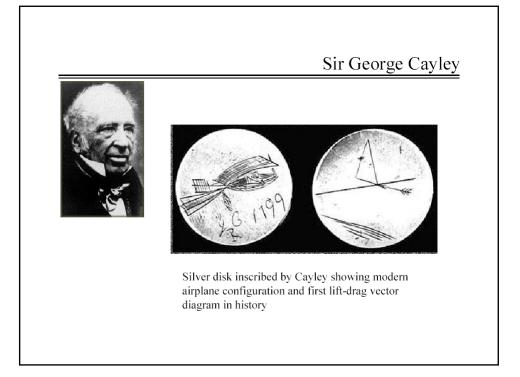




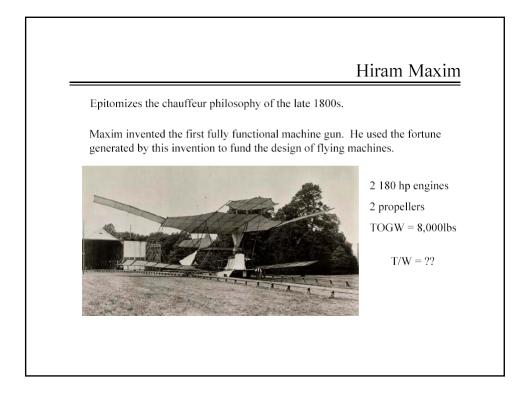


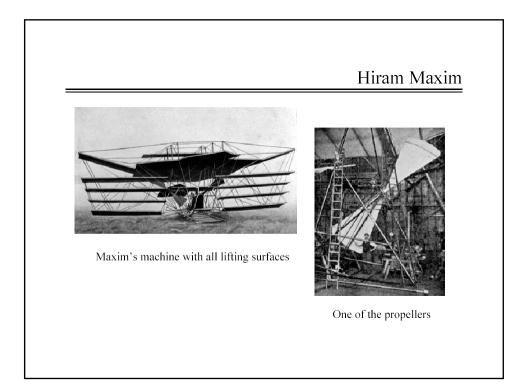






"Chauf	feurs
19th century flying machine inventors who were obsessed with "brute for given enough thrust (or horsepower) from an engine, the airplane could forced into the air.	
circular argument:	
more power \longrightarrow more weight \longrightarrow must move faster to gener	
controllability once it was in the air.	gni to
The failure of this philosophy led to the idea of developing engines w more power but less weight.	ith
T/W ratio is now a critical parameter in aircraft design.	





"Airman's Approach"

Philosophy that, in order to design a successful flying machine, it is necessary to fly unencumbered by a power plant-learn to fly before you put an engine on the aircraft.

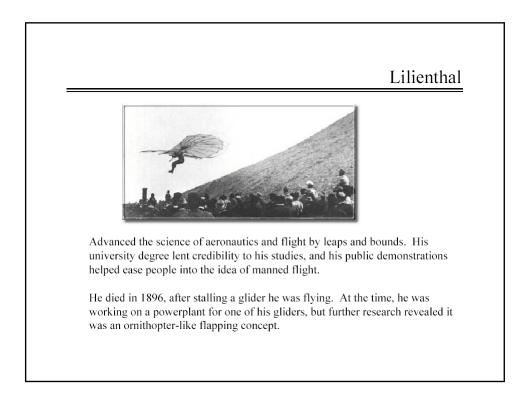
Antithesis of the chauffeur philosophy

Otto Lilienthal, a German mechanical engineer, designed and flew the first successful gliders in history.

clearly demonstrated the aerodynamic superiority of cambered (curved) airfoils in comparison to flat, straight surfaces

Lilienthal's book *Bird Flight as the Basis of Aviation* was far and away the most important and definitive contribution to the budding science of aerodynamics in the 19th century.

Published first drag polars ever.



Samuel P. Langley

Was contracted to build a flying machine for the U.S. government.

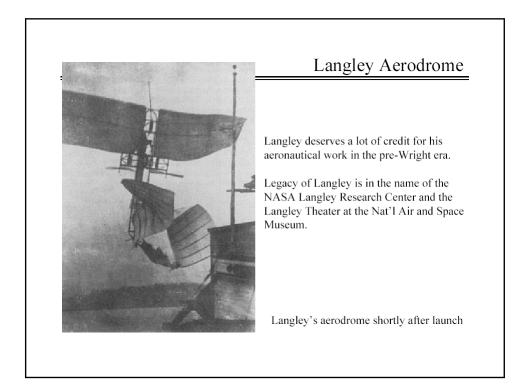
Began a series of aerodynamic experiments in 1887, using a whirling arm and published a book *Experiments in Aerodynamics* in 1890.

Was third Secretary of the Smithsonian Institution

Successful in flying several small scale, unmanned, powered aircraft, which he called *aerodromes*. These were the first steam-powered, heavier-than-air machines to successfully fly.

Along with Charles Manly designed and developed the best airplane powerplant until the beginning of World War I (124lb, 52.4 hp). T/W=???

Langley's attempt to build a manned aerodrome failed but had a superb powerplant, marginally good aerodynamics, but was structurally unsound. Launched and crashed on Oct 7 and December 8, 1903



Era of Strut and Wire Biplanes

The Wright Brothers

The 1903 Wright Flyer ushered in the era of successful strut-and-wire biplanes- an era that covers the general period of 1903 to 1930.

Unlike Langley's full-scale aerodrome, there were no fatal "weak links" in the design of the Wright Flyer-they were the first true aeronautical engineers.

In the 1903 Wright Flyer they had gotten it all right-the propulsion, aerodynamic, structural, and control aspects were carefully calculated and accounted for during its design

Propulsion was achieved by a four-cylinder in-line engine designed and built by Orville Wright. It produced close to 12 hp and weighed 140 lbs. T/W = ??

The engine drove two propellers via a bicycle-like chain loop.

	The Wright Brothers
was th nothir	ropellers were masterpieces of aerodynamic design. Wilbur Wright he first person to recognize the fundamental principle that a propeller is ng more than a twisted wing. He conceived the first viable propeller and blade element theory.
	Prop efficiency of 70% Langley's prop efficiencies were on the order of 52% What are modern prop efficiencies?
its suc aroun	ontrol features of the Wright Flyer are also one of the basic reasons for cess. They were the first to recognize the importance of flight control d all three axes of the airplane: pitch control, yaw control, roll control. were the first to appreciate the value of roll control and used wing ng.
	ructural features were patterned partly after the work of Chanute and own experience designing bicycles-another example of evolutionary

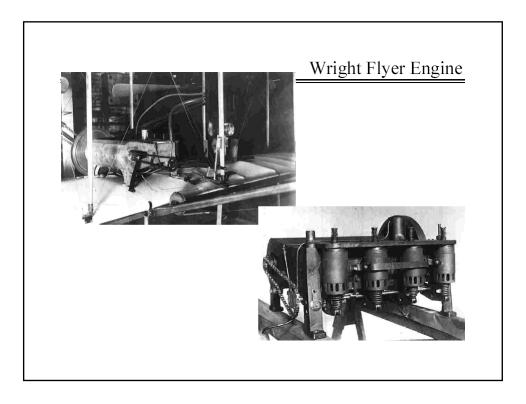
Wright Brothers-Aerodynamics

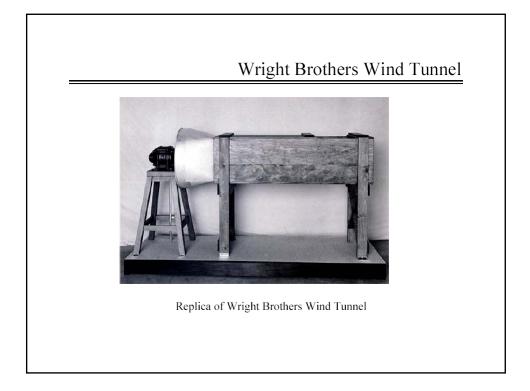
The Wrights were well aware that the major measure of aerodynamic efficiency is the lift to drag ratio, L/D, and did three things:

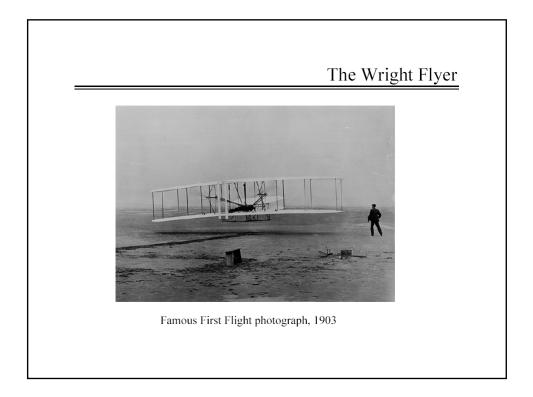
Chose an airfoil shape that, based on the collective data from their wind tunnel tests, would give a high L/D (on the order of 6)

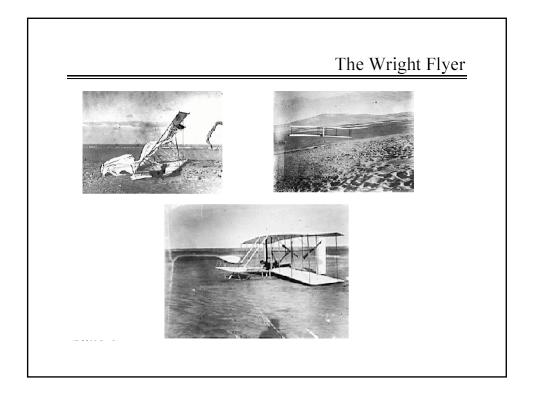
Chose an aspect ratio of 6 for the wings, although they didn't know that by increasing the aspect ratio from 3 to 6, they reduced the induced drag by a factor of 2 (Prandtl showed this in Germany 15 years later).

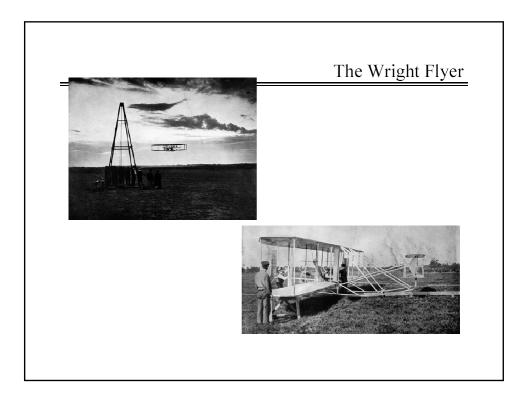
Very conscious of the importance of parasite drag, which in their day was called head resistance (used empirical formulas obtained from Octave Chanute in his book Progress in Flying Machines). To combat drag, they decided to fly prone (laying down). They did not, however, realize the high drag caused by the wires.

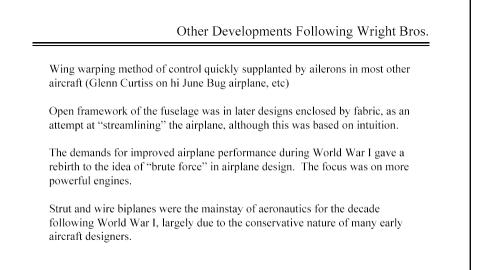






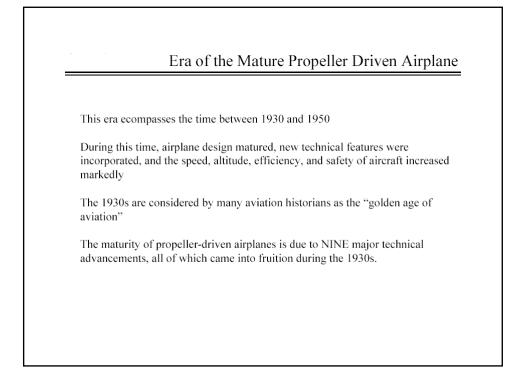






The famous French airplane designer, Louis Brequet, however, appreciated the need to bring to a minimum the value of D/L and even called for retractable landing gear in 1924.

Sponsorship of air races, such as the Schneider Cup races, resulted in specifically designed racing airplanes and they substantially contributed to the demise of the era of strut and wire biplanes and the beginning of mature, propeller driven aircraft.



9 Major Technical Advancement	
FIR	ST- Cantilevered-Wing Monoplane:
	Bleriot first crossing of the English Channel (1909) in a monoplane caused a surge of popularity; however, an inordinate number of crashes precipitated by structural failure of the monoplane wings (1913) helped to reinforce distrust.
	Hugo Junkers designed and built the first all steel cantilever monoplane (1915).
	Ford Trimotor first widely accepted monoplane in U.S. (1926). Rockne killed.
	Boeing Monomail (1930) embodied two technical developments: all metal stressed skin construction retractable landing gear

SECOND- Retractable Landing Gear	
THIRD- Cowli	ng for Radial Piston Engines:
Clark de XPS-1	signed a primitive cowling in 1922 for the Dayton-Wright
	Townsend designed a ring that was installed on the Boeing ail (noticable decrease in drag with no interference with ooling)
through understa	reakthrough in engine cowlings was due to NACA in U.S. a systematic series of wind tunnel tests with the objective of nding the aerodynamics of engine cowlings and designing an shape for them.
	esults indicated 60% reduction with full cowling and by esign of the cowling enhanced cooling of the engine could be

	9 Major Technical Advancements
FOURTH- Variable Pi	itch Propeller
	the twist is designed so that each airfoil section is at its attack to the relative airflow, usually approximately to he airfoil.
	itch propeller is operating at maximum efficiency at ed; a tremendous disadvantage of a fixed pitch
conditions is not an mechanical device	f the propeller during flight to operate at near-optimum a easy mechanical task. The first practical and reliable for varying propeller pitch was designed in 1933. ort used variable pitch propeller to improve its ficiency.
Constant speed pro	pellers debuted in the 1930s

9 Major Technical Advancements

FIFTH- High Octane Aviation Fuel

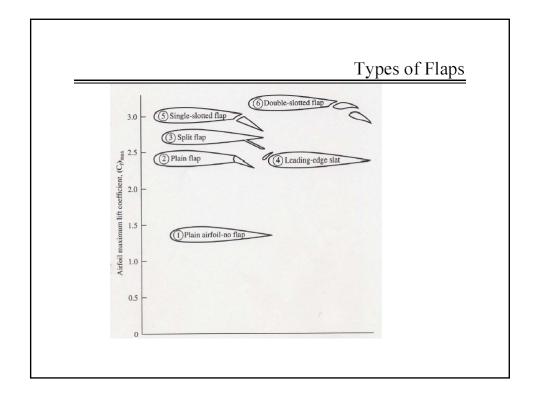
Engine pinging caused permature ingition as early as 1911

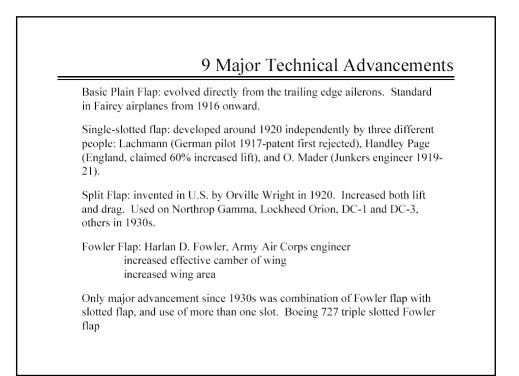
Tetraethyl lead found to reduce engine knicking- GM and Standard Oil formed new company, Ethyl Gasoline Corporation, to produce "ethyl" gas with lead

Army Air Corps (1930) adopted 87 octane gas as its standard fuel; in 1935 raised to 100 octane. 100 octane allowed much higher compression ratios inside the cylinder head, and hence more power for engine.

Introduction of 100 octane fuel, as well as other technological improvements, allowed Curtiss-Wright to increase power of its R-1820 Cyclone engine from 500 to 1,200 hp in the 1930s.

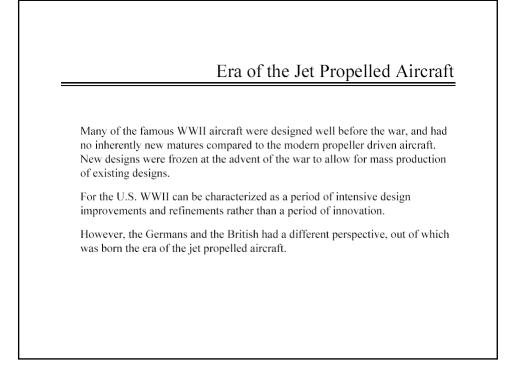
	9 Major Tec	chnical Advancements
SIXTH- Developm	ent of High Lift Devices	
0	criterion dictates the ratio of	e sufficient lift at takeoff and f airplane weight to wing area
differences res		ther speed causes larger pressure Therefore, high lift devices a sized for cruise.

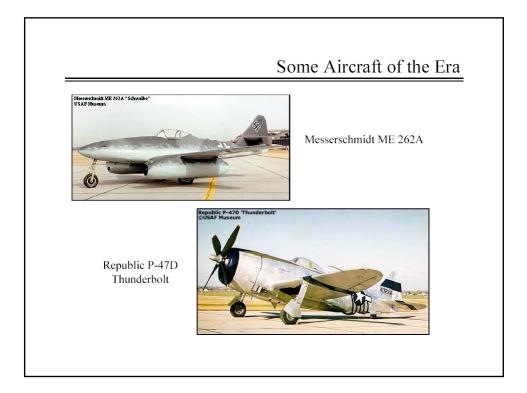




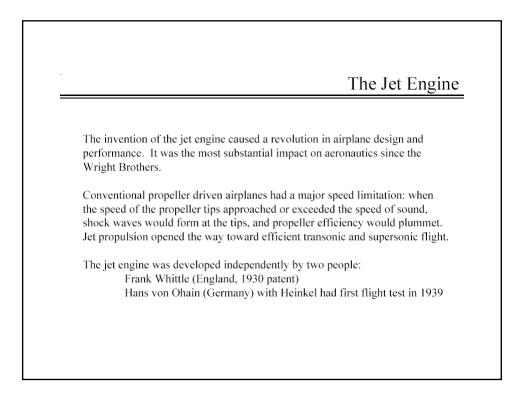
Description of the provided of the provided

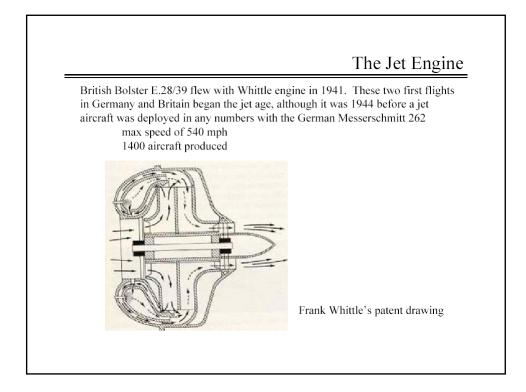
NINTH- Laminar Flow Airfoils	
layer (Pra and fusela NACA de almost 50	s of flow: laminar flow and turbulent flow in the boundary ndtl 1904). About 99% of the boundary layer along wings age in flight is turbulent, creating high skin friction drag. veloped laminar flow airfoils which reduced airfoil drag by %. However, in practice laminar flow wing never payed off duction (manufacturing imperfections, bugs, etc).
They had	low airfoils turned out to be very good on high speed airfoils. a much higher critical Mach number and delayed onset of ibility problems encountered by many high-speed airplanes in 1940s.



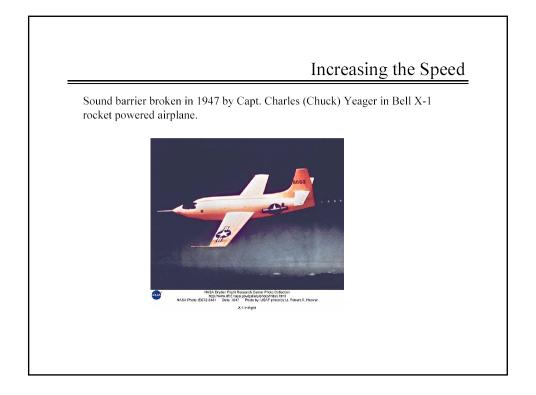


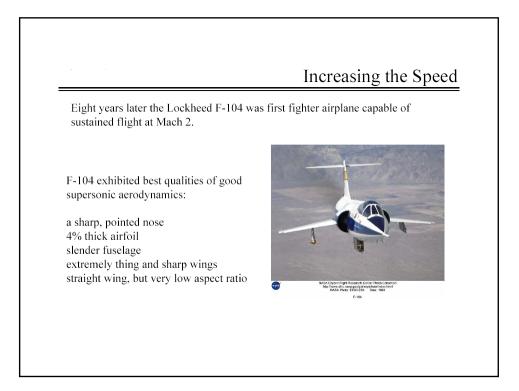


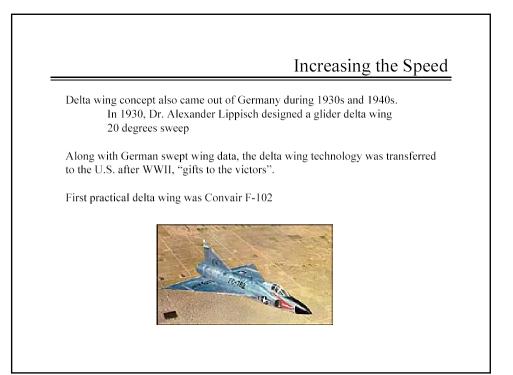


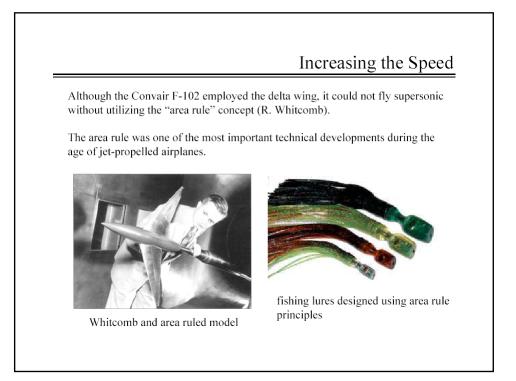


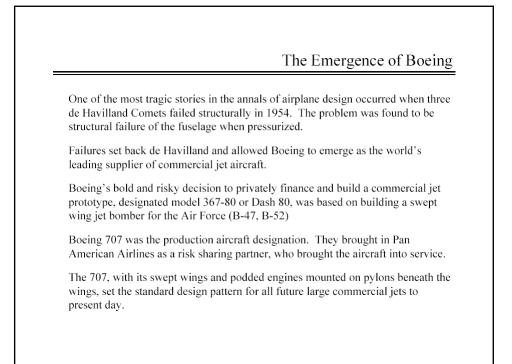
	Increasing the Speed
	e era of jet propelled aircraft is characterized by a number of design features que to airplanes intended to fly at, near, or beyond the speed of sound.
Swept	Wing
	subsonic - fly closer to the speed of sound, increased the airplane's critical Mach number
	supersonic - can keep wing's leading edge inside the Mach cone from the nose of the fuselage
	for high subsonic or supersonic, and airplane with a swept wing can fly faster than one with a straight wing, everything else being equal
	first introduced in a public forum in 1935 by Adolf Bussman, German aerodynamicist. Later classified by German Luftwaffe as a military secret
	R.T. Jones, NACA aerodynamicist, worked out swept wing theory toward end of WWII
	North American Aircraft F-86 world's first successful operational swept wing aircraft in 1948.

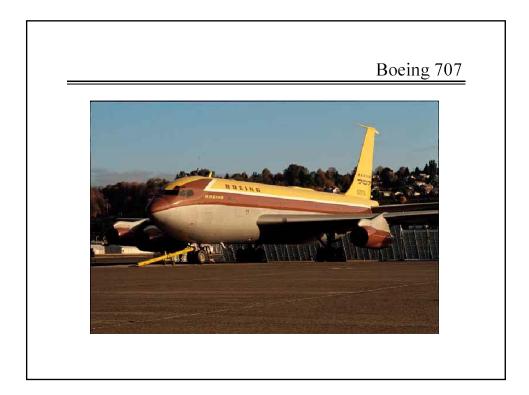










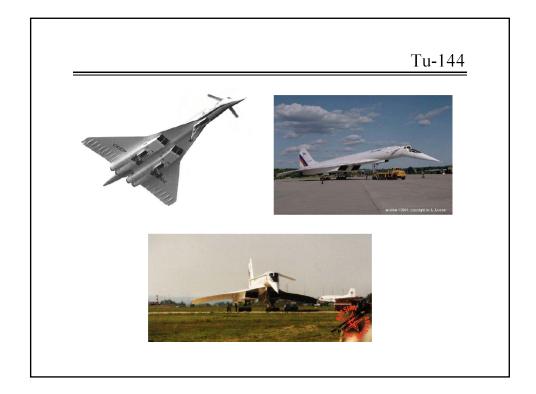


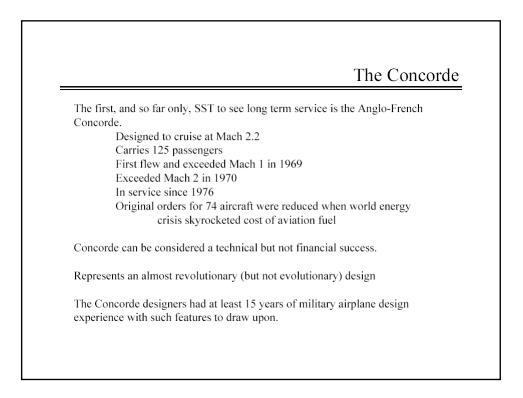
Boeing 747

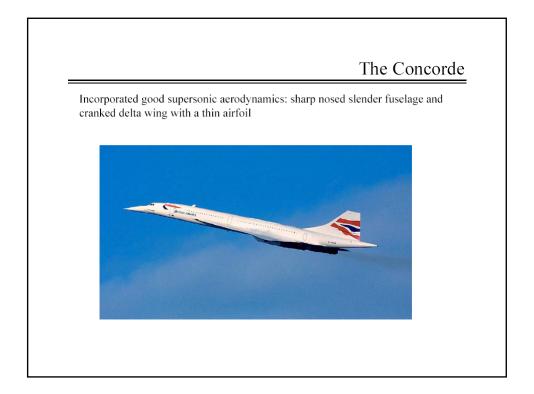
Boeing's second bold move was to "go for the big one" in 1966 after losing to Lockheed in the Air Force C-5 competition. Boeing 747 first flew in 1969 and entered service with Pan Am in 1970.

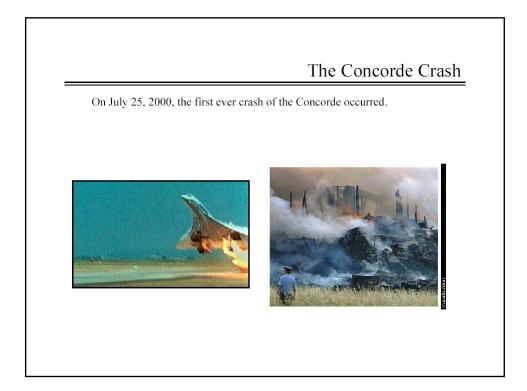


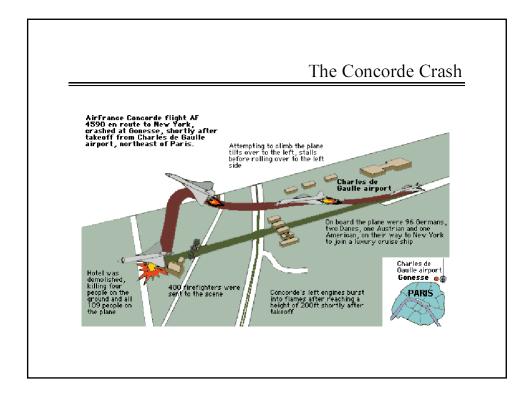
	Supersonic Transports (SSTs)
In the 1960s, the U.S., England	, and France began development of SSTs.
The Tu-144 first flew on Dec 3 built, but none entered extended	shed an SST design into production and service. 1, 1968. More than a dozen of these aircraft were d service, presumably due to unspecified stroyed in a dramatic accident at the 1973 Paris
2707 was the winner in 1966. I variable sweep wing was pursu- caught in an upward spiral of in	ted a design competition for an SST. The Boeing Design turned into a nightmare for Boeing. First a ed and then junked, then the new design was acreased weight and development cost. When at cost hit about \$5billion, Congress terminated the

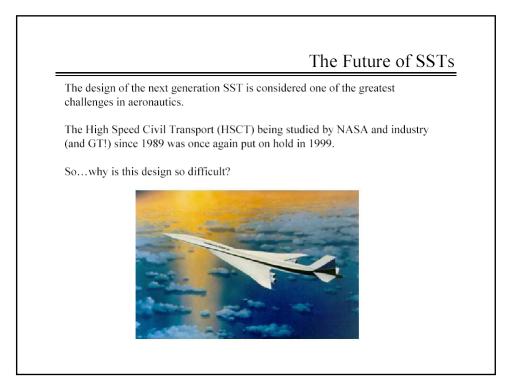




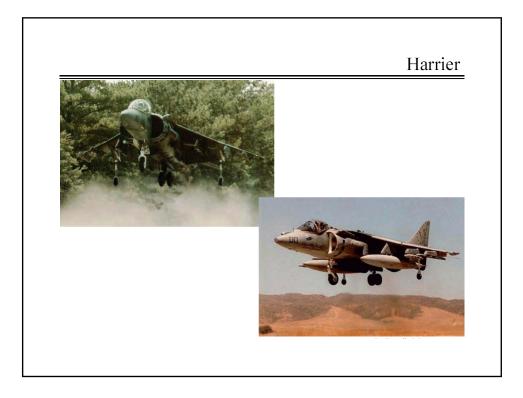


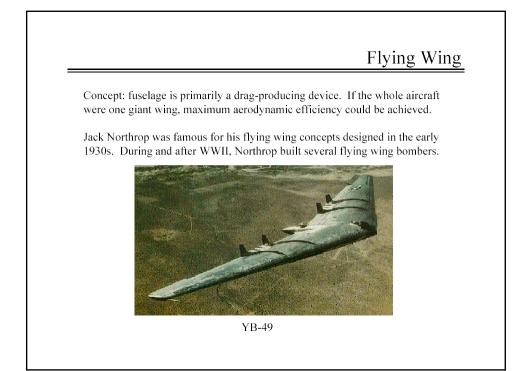


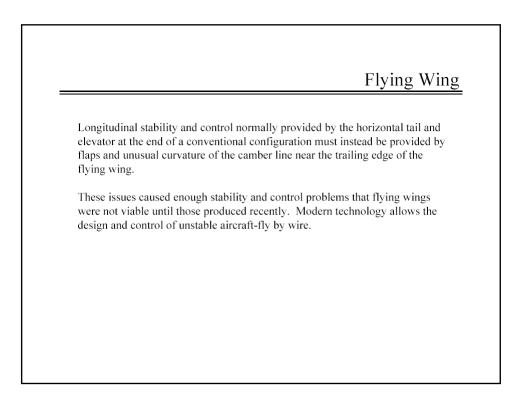




<text><text><text><text><text><text>







Modern Flying Wing-B2 Bomber



